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PIONEER 10 72-012A-00H

PIONEER 11 73-019A-00G

TRAJECTORY DATA - SFDU FORMAT

PIONEER 10

72-012A-00H

## TRAJECTORY DATA - SFDU FORMAT

This data set consists of 2 magnetic tapes. The tapes were written on 9-track, 6250 bpi, in SFDU format, and can be read with the VAX COPY command in VMS directory format. The first two files on the tape contains the volume description and the file format. These are followed by the data files. All data records are of the same length, 512 ASCII characters or bytes per physical record. The C tapes are 3480 bpi. The D and C numbers and time span are as follows:

D#	C#	FILES	TIME SPAN		
D-085883	C-029141	39	04/06/73 - 01/01/1990		
D-108359	C-032396	23	01/01/90 - 01/01/2000		

## PIONEER 11

73-019A-00G

# TRAJECTORY DATA - SFDU FORMAT

This data set consists of 2 magnetic tapes. The tapes were written on 9-track, 6250 bpi, in SFDU format, and can be read with the VAX COPY command in VMS directory format. The first two files on the tape contains the volume description and the file format. These are followed by the data files. All data records are of the same length, 512 ASCII characters or bytes per physical record. The C tapes are 3480 bpi. The D and C numbers and time span are as follows:

D#	C#	FILES	TIME SPAN		
D-085884	C-029142	37	04/06/73 - 01/01/1990		
D-108360	C-032397	23	01/01/90 - 01/01/2000		

### CCSD3ZF0000100000001CCSD3VS00002MRK\*\*\*001

USA\_NASA\_NSSD\_P10Z\_0002 VOL\_IDENT:

VOL\_CREATION\_DATE: 1995-07-11

. . MEDIUM\_DESCRIPTION: 1/2-inch, 9-track, 6250 bpi magnetic tape

David W. Lozier TECHNICAL\_CONTACT:

NASA Ames Research Center (Mail Stop 244-14)

Moffett Field, CA 94035

David\_Lozier@QMGATE.ARC.NASA.GOV Electronic Mail:

415-604-3147 Telephone:

PREV\_VOLS: USA\_NASA\_NSSD\_P10Z\_0001

CCSD\$\$MARKERMRK\*\*001CCSD3SS00002MRK\*\*002

PIONEER 10 TRAJECTORY EPHEMERIS ARCHIVE (Extension to DATA\_SET\_NAME:

2000)

Pioneer 10 trajectory data were generated using the DATA\_SOURCE: Jet Propulsion Laboratory's (JPL) "Double Precision

Trajectory" (DPTRAJ) computer program, an element

of the JPL Navigation Software System, reference (A).

DPTRAJ creates a spacecraft ephemeris by integrating the equations of motion for a full set of acceleration models. For the Pioneer 10 trajectory aphemeris, DPTRA. integration was limited to the N-body gravitational terms (planets, Sun and Moon) with Earth oblateness, Jupiter oblateness (J2 and J4), and a simple solar pressure model included.

Initial conditions for the DPTRAJ runs were obtained from various Orbit Determination (DD) solutions performed during the mission. A set of seven spacecraft initial state vectors (position, velocity and epoch) was compiled to cover all phases of the mission from launch to 1 January 1990 (Volume USA\_NASA\_NSSD\_P10Z\_0001).

- (1) Launch to second midcourse maneuver
- (2) Second midcourse maneuver to first "IO" targeting maneuver
- (3) First "IO" targeting maneuver to second "IO" targeting maneuver
- (4) Second "IO" targeting maneuver to Jupiter phase
- (5) Jupiter encounter DD
- (6) Jupiter encounter to 1983 solar escape
- (7) 1983 solar escape to 1 January 1990

A complete set of input for the seven DPTRAJ runs is provided in reference (8).

This volume extends the Pioneer 10 trajectory from 1 January 1990 to 1 January 2000.

The DPTRAJ input for volume USA\_NASA\_NSSD\_P10Z\_0002 is provided in reference (D).

SCIENTIFIC\_CONTACT:

SPACECRAFT\_CHARACTERISTICS: Pioneer 10 is a 550 pound spin stabilized

none

spacecraft powered by four RTGs. Communications with the Earth is provided by a 9 ft. diameter parabolic high gain antenna located on the spin axis of the spacecraft. Periodic precession maneuvers are performed to maintain

Sarth pointing of the high gain antenna. The spacecraft

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1/1/1990 - 1/1/2000

FATAR CONTROL CARDS

6514421 ANALYZE BLP, LABELS=NO, PRTLEN=2000 PRINT LF=1, B=1-29 PRINT LF=2, B=1-47 PRINT LF=3, B=1-5 PRINT LF=22, B=15-24 PRINT LF=23, B=1-4

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE TAPE TO BE UNIT SERIAL DEN TRTCH 5B5 CMS20 38000 ANALYZED

# FATAR DETAIL REPORT

BLOCK LNGTH/ NUMBER DISPL ж ж +00080 +00160 +00240 +00320 +00160 +00240 +00400 +00320 +00080 \* \* 460 495 PRINT REQUESTED PRINT REQUESTED START FILE MESSAGE/ BLOCK TYPE gram, an element of the JPL Navigation Software System, reference (A).

DPTRAJ creates a spacecraft ephemeris by integrating generated using the
) "Double Precision + .5...10...15...20...25...30...35...40...45...50...55...60...65...70...75...80 (COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA) DATA\_SET\_NAME: PIONEER 10 TRAJECTORY EPHEMERIS ARCHIVE (Extension to 2000)

DATA\_SOURCE: Pioneer 10 trajectory data were Jet Propulsion Laboratory's (JPL Precision + Trajectory'' (DPTRAJ) computer pro CCSD3ZF0000100000001CCSD3VS00002MRK\*\*001 VOL IDENT: USA\_NASA\_NSSD 102 VOL\_CREATION\_DATE: 1995-07-11 MEDIUM\_DESCRIPTION: 1/2-inch, 9-t 150 bpi magnetic tape TECHNICAL\_CONTACT: David W. Lozier NASA Am 11 Nasa Am OCSD\$\$MARKERMRK\*\*001CCSD3SS00

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PRINT REQUESTED REQUESTED For the Pioneer 10 trajectory ephemeris, DPTRAJ integration o the N-body gravitational terms (planets, Sun and Moon) w l state vectors (position, velocity and epoch) was hases of the mission from launch to 1 January 1990 USA\_NASA\_NSSD\_P10Z\_0001). model included. eness, models.

"""" presumeris, DPTRAJ integration was limited to the solution with Earth oblate included.

Initial conditions for the DPTRAJ runs were obtained from various Orbit Determination (OD) solutions performed during the mission. A set of seven spacecraft epoch) was compiled to 1 January 1990 (Volume cover all (1) Launch initia σ

+00080 +00160 +00240 +00320

+00080 +00160 +00240

+00400

454

+00320

+00400

455

+00080 +00160 +00240 PRINT REQUESTED targeting maneuver to (5) Jupiter encounter Jupiter encounter OD second "IO" targeting maneuver (3) First "IO" targeting Jupiter phase (6) Jupiter encounter to 1 maneuver Second "IO"

neuver

to second midcourse maneuver neuver to first "IO"

targeting maneuver

Second midcourse ma

FATS070 CONTROL CARD TABLE SIZE IS FAST ANALISIS OF TAPE AND RECOVERY --4096 BYTES FATAR VER 4.3.5 INNOVATION DATA PROCESSING AUTHORIZED 10/04/95

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ca/06/13- 01/04/90

**FATAR CONTROL CARDS** 

2-1 ANALYZE BLP, LABELS=NO, PRTLEN=9000 PRINT LF=ALL, B=1-5, CHAR

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE UNIT SERIAL 5B2 CMS22 DEN TRTCH ANALYZED

ж ж ж BL OCK ж ж \* \* \* ж ж ж N DISPL ж ж \* \* \* \* ж ж 80 80 80 80 START FILE END OF FILE PRINT REQUESTED PRINT REQUESTED START FILE MESSAGE/ BLOCK TYPE PRINT REQUESTED PRINT REQUESTED FATS038 DATA FILE IN ISCII/ASCII - TRANSLATED TO EBCDIC N ļ FILE CONTAINED 1...5...10...15...20...25...30...35...40...45...50...55...60...65...70...75...80 (COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA) HDR2D0051200512 HDR1VOLDESC.SFD VOLITRJP10 **FATAR DETAIL REPORT BLOCKS** TRJP1000010001000100 92056 92056 000000DECFILE11A D%B44440010011

W

FATS038 DATA FILE IN ISCII/ASCII - TRANSLATED TO EBCDIC

+00160 +00240 +00320 +00080 +00160 +00240 +00320 +00080 +00400 +00480 +00400 512 PRINT REQUESTED PRINT PRINT REQUESTED REQUESTED 0063 models. For the Pioneer 10 trajectory ephemeris, DPTRAJ0059 integration was limited to the N-body gravitational 0059 terms (planets, Sun and Moon) with Earth oblateness,0058 Jupiter oblateness (J2 and J4), and a simple solar 0032 pressure model included. 00040059 Initial conditions for the DPTRAJ runs were obtained0055 from various Orbit Determination (OD) solutions 0063 perf 0059DATA\_SET\_NAME: PIONEER 10 TRAJECTORY EPHEMERIS ARCHIVE0070DATA\_SOURCE: neer 10 trajectory data were generated using the 0079
Propulsion Laboratory's (JPL) "Double Precision0078
ectory" (DPTRAJ) computer program, an element 0060 of the JPL Navigation (are System, reference (A).00040059 DPTRAJ creates a spacecraft ephemeris in the system of the JPL Navigation (are System). 0044CCSD3ZF000010000001CCSD3VS00002MRK\*\*00100040038VOL\_IDENT: USA\_NASA\_NSSD\_P10
Z\_00010033VOL\_CREATION\_DATE: 1991-12-060065MEDIUM\_DESCRIPTION: 1/2-inch, 9-track
, 6250 bpi magnetic tape0038TECHNICAL\_CONTACT: David W. Lozier0051 NASA Ames R
esearch Center (Mail Stop 244-14)0033 Moffett Field, CA 940350007 0040
Electronic Mail: Span OPS::LOZIER0026 NASAMAIL/DLOZIER0031 Telephone: 41
5-604-31470022 FTS 464-31470020PREV\_VOLS: none00040044CCSD\$\$MARKERMRK\*\*\*001
CCSD3SS000002MRK\*\*\*0020004----tegrating0061 erence (A).00040059 DPTRAJ creates a spacecraft ephemeris by in the equations of motion for a full set of acceleration Softw Jet

+00240 +00320 +00080

FATS070 CONTROL CARD TABLE SIZE IS FAST ANAL.SIS OF TAPE AND RECOVERY --4096 BYTES FATAR VER 4.3.5 INNOVATION DATA PROCESSING AUTHORIZED 10/04/95

2--ANALYZE BLP, LABELS=NO, PRTLEN=9000 PRINT LF=ALL, B=1-5, CHAR **FATAR CONTROL CARDS** 

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE TAPE TO BE UNIT SERIAL DEN TRTCH 5B2 CMS23 38000 ANALYZED

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URIAL

FATAR DETAIL REPORT

ж ж BLOCK ж \* \* ж ж 4 ú N LNGTH/ DISPL ж ж \* \* ж ж \* \* 80 80 80 BLOCK TYPE START FILE START FILE PRINT REQUESTED PRINT REQUESTED PRINT REQUESTED END OF FILE PRINT REQUESTED FATS038 DATA FILE IN ISCII/ASCII - TRANSLATED TO EBCDIC MESSAGE/ N į FILE CONTAINED HDR1VOLDESC.SFD HDR2D0051200512 VOL1TRJP11 5...10...15...20...25...30...35...40...45...50...55...60...65...70...75...80 (COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA) Þ BLOCKS TRJP1100010001000100 92057 92057 000000DECFILE11A D%B44440010011

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FATS038 DATA FILE IN ISCII/ASCII - TRANSLATED TO EBCDIC

3 512 +00080 +00160 +00240 +00320 2 +00080 +00160 +00240 +00080 +00160 +00240 +00400 +00320 +00480 +00400 +00320 +00480 PRINT REQUESTED PRINT REQUESTED PRINT REQUESTED 0082 ional0080 on0084 **DPTRAJ0079** the equations of motion for a full set of acceleration models. For the Pioneer II trajectory ephemeris, integration was limited to the N-body gravitat terms (planets, Sun and Moon) with Earth oblate Jupiter oblateness (J2 and J4), Saturn oblatene of the Iraje Jet Pio P.

**FATAR CONTROL CARDS** 

311 ANALYZE BLP, LABELS=NO, PRTLEN=2000 PRINT LF=1, B=1-31 PRINT LF=2, B=1-47 PRINT LF=3, B=1-5 PRINT LF=22, B=15-24 PRINT LF=23, B=1-4

6 Span /1/40 - 1/1/2000

165-50-29

6---5---

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE TAPE TO BE ANALYZED UNIT SERIAL DEN TRICH 5B5 CMS21 38000

# **FATAR DETAIL REPORT**

4 461 +00080 +00160 +00240 +00220 +00320	3 466 +00080 +00160 +00240 +00320 +00400	2 +00080 +00160 +00160 +00240 +00320 +00400 +00480	1 +00080 +00160 +00240 +00240 +00320 +00400 +00480	* * * * * * *	BLOCK LNGTH/ NUMBER DISPL
PRINT REQUESTED	PRINT REQUESTED	PRINT REQUESTED	PRINT REQUESTED	START FILE 1	MESSAGE/ BLOCK TYPE
Initial conditions for the DPTRAJ runs were obta ined.  From various Orbit Determination (OD) solutions or the mission. A set of eleven of eleven of the mission. A set of eleven or the mission, velocity and epoch) was computed to cover all phases of the mission from launch to 1 January 1990 (volume of the USA_NASA_NSSD_P11Z_0001).	K the equations of motion for a full set of accele ration models. For the Pioneer Il trajectory ephemer is, DPTRAJ   integration was limited to the N-body gravitational & terms (planets, Sun and Moon) with Earth ob lateness, Jupiter oblateness (J2 and J4), Saturn oblateness (J2), and a simple solar pressure model included.	4 DATA_SET_NAME: PIONEER 11 TRAJECTORY EPHEMERIS ARCHIVE (Extension to 2000) DATA_SOURCE: Pioneer 11 trajectory data were generated using the   2000) DATA_SOURCE: Pioneer 11 trajectory data were Jet Propulsion Laboratory's (JPL) "Double Precision (	CCSD3ZF0000100000001CCSD3VS00002MRK************************************		15101520253035404550556065707580 (COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA)

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PRINT REQUESTED

midcourse maneuver ¢

(2) First midcourse maneuver to second

(3) Se